

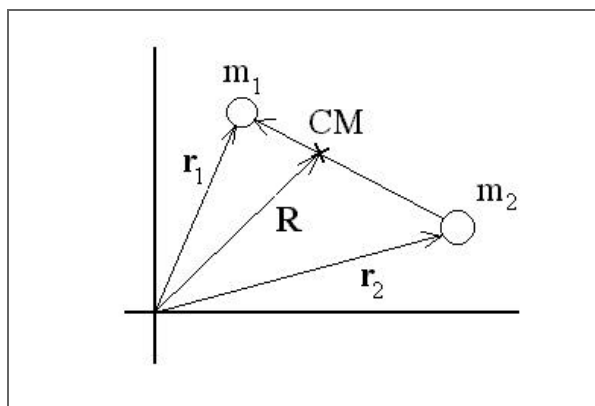
Motion in a Central-Force Field

1 Motion in the Center-of-Mass Frame

The Lagrangian for an isolated two-particle system is

$$L = \frac{m_1}{2} |\dot{\mathbf{r}}_1|^2 + \frac{m_2}{2} |\dot{\mathbf{r}}_2|^2 - U(\mathbf{r}_1 - \mathbf{r}_2),$$

where \mathbf{r}_1 and \mathbf{r}_2 represent the positions of the particles of mass m_1 and m_2 , respectively, and $U(\mathbf{r}_1, \mathbf{r}_2) = U(|\mathbf{r}_1 - \mathbf{r}_2|)$ is the potential energy for an isolated two-particle system (see Figure below).



Let us now define the position \mathbf{R} of the center-of-mass (CM)

$$\mathbf{R} = \frac{m_1 \mathbf{r}_1 + m_2 \mathbf{r}_2}{m_1 + m_2},$$

and define the inter-particle vector $\mathbf{r} = \mathbf{r}_1 - \mathbf{r}_2$, so that the particle positions can be expressed as

$$\mathbf{r}_1 = \mathbf{R} + \frac{m_2}{M} \mathbf{r} \quad \text{and} \quad \mathbf{r}_2 = \mathbf{R} - \frac{m_1}{M} \mathbf{r},$$

where $M = m_1 + m_2$ is the total mass of the two-particle system. The Lagrangian of the isolated two-particle system thus becomes

$$L = \frac{M}{2} |\dot{\mathbf{R}}|^2 + \frac{\mu}{2} |\dot{\mathbf{r}}|^2 - U(\mathbf{r}),$$

where

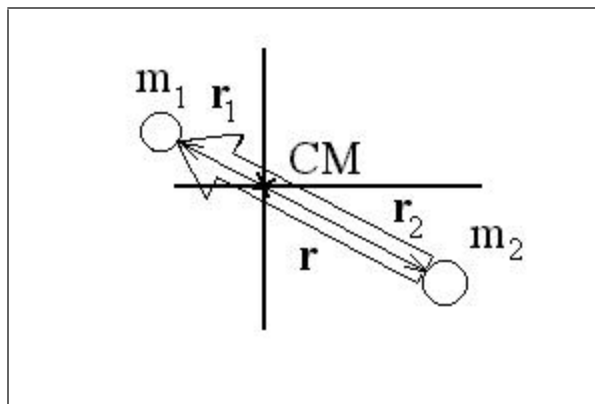
$$\mu = \frac{m_1 m_2}{m_1 + m_2} = \left(\frac{1}{m_1} + \frac{1}{m_2} \right)^{-1}$$

denotes the *reduced* mass of the two-particle system.

For an isolated system, the CM canonical momentum

$$\mathbf{P} = \frac{\partial L}{\partial \dot{\mathbf{R}}} = M \dot{\mathbf{R}}$$

is a constant of the motion. The CM reference frame is defined by the condition $\mathbf{R} = 0$, i.e., we move the origin of our coordinate system to the CM position (the Figure below shows the case where $m_1 > m_2$).



In this case, the Lagrangian for an isolated two-particle system in the CM reference frame is

$$L(\mathbf{r}, \dot{\mathbf{r}}) = \frac{\mu}{2} |\dot{\mathbf{r}}|^2 - U(\mathbf{r}),$$

where

$$\mathbf{r}_1 = \frac{m_2}{M} \mathbf{r} \quad \text{and} \quad \mathbf{r}_2 = -\frac{m_1}{M} \mathbf{r}.$$

Hence, once the Euler-Lagrange equation for \mathbf{r}

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\mathbf{r}}} \right) = \frac{\partial L}{\partial \mathbf{r}} \quad \rightarrow \quad \mu \ddot{\mathbf{r}} = -\nabla U(\mathbf{r})$$

is solved for $\mathbf{r} = \mathbf{r}(t)$, the motion of m_1 and m_2 are determined through $\mathbf{r}_1(t) = (m_2/M) \mathbf{r}(t)$ and $\mathbf{r}_2(t) = -(m_1/M) \mathbf{r}(t)$.

2 Motion in a Central-Force Field

A particle moves under the influence of a central-force field $\mathbf{F}(\mathbf{r}) = F(r) \hat{\mathbf{r}}$ if the force on the particle is independent of the angular position of the particle about the center of force

and depends only on its distance r from the center of force. Here, the magnitude $F(r)$ (which is positive for a repulsive force and negative for an attractive force) is defined in terms of the central potential $U(r)$ as $F(r) = -U'(r)$.

2.1 Lagrangian Formalism

The motion of two particles in an isolated system takes place on a two-dimensional plane. When these particles move in a central-force field, the Lagrangian is simply

$$L = \frac{\mu}{2} (\dot{r}^2 + r^2 \dot{\theta}^2) - U(r), \quad (1)$$

where polar coordinates (r, θ) are most conveniently used. Since the potential U is independent of θ , the canonical momentum

$$p_\theta = \frac{\partial L}{\partial \dot{\theta}} = \mu r^2 \dot{\theta} \equiv \ell \quad (2)$$

is a constant of motion (here, labeled ℓ). The Euler-Lagrange equation for r , therefore, becomes the radial force equation

$$\mu (\ddot{r} - r \dot{\theta}^2) = \mu \ddot{r} - \frac{\ell^2}{\mu r^3} = F(r). \quad (3)$$

In this description, the planar orbit is parametrized by time, i.e., once $r(t)$ and $\theta(t)$ are obtained, a path $r(\theta)$ onto the plane is defined.

Since $\dot{\theta}$ does not change sign on its path along the orbit, we may replace \dot{r} and \ddot{r} with $r'(\theta)$ and $r''(\theta)$ as follows. First,

$$\dot{r} = \dot{\theta} r' = \frac{\ell r'}{\mu r^2} = -\frac{\ell}{\mu} \left(\frac{1}{r}\right)'$$

Next, using Eq. (2)

$$\ddot{\theta} = -\frac{2\ell \dot{r}}{\mu r^3} = -\frac{2\ell^2 r'}{\mu^2 r^5},$$

we find

$$\ddot{r} = \ddot{\theta} r' + \dot{\theta}^2 r'' = -\frac{\ell^2}{\mu^2 r^5} [2(r')^2 - r r'']$$

Lastly, using the identity

$$\left(\frac{1}{r}\right)'' = \left(-\frac{r'}{r^2}\right)' = \frac{1}{r^3} [2(r')^2 - r r''],$$

we find an expression for \ddot{r} :

$$\ddot{r} = -\frac{\ell^2}{\mu^2 r^2} \left(\frac{1}{r}\right)''$$

and the radial force equation (3) becomes

$$s'' + s = -\frac{\mu}{\ell^2 s^2} F(1/s) = -\frac{d\bar{U}(s)}{ds}, \quad (4)$$

where $s(\theta) = 1/r(\theta)$ and $\bar{U}(s) = (\mu/\ell^2) U(1/s)$.

Note that the form of the potential can be calculated from the solution $s(\theta) = 1/r(\theta)$ as follows. For example, consider the particle trajectory described in terms of the solution $r(\theta) = r_0 \sec(\alpha\theta)$, where r_0 and α are constants, then

$$s'' + s = -(\alpha^2 - 1) s = -\frac{d\bar{U}(s)}{ds},$$

and thus

$$\bar{U}(s) = \frac{1}{2} (\alpha^2 - 1) s^2 \rightarrow U(r) = \frac{\ell^2}{2\mu r^2} (\alpha^2 - 1).$$

Note also that the function $\theta(t)$ is determined from the relation

$$\dot{\theta} = \frac{\ell}{\mu r^2(\theta)} \rightarrow t(\theta) = \frac{\mu}{\ell} \int_0^\theta r^2(\theta) d\theta.$$

Returning to our example, we find

$$t(\theta) = \frac{\mu r_0^2}{\alpha \ell} \int_0^{\alpha\theta} \sec^2 \phi d\phi = \frac{\mu r_0^2}{\alpha \ell} \tan(\alpha\theta) \rightarrow r(t) = r_0 \sqrt{1 + \left(\frac{\alpha \ell t}{\mu r_0^2}\right)^2}$$

and the total energy

$$E = \frac{\alpha^2 \ell^2}{2\mu r_0^2},$$

is determined from the initial conditions $r(0) = r_0$ and $\dot{r}(0) = 0$.

2.2 Hamiltonian Formalism

The Hamiltonian for the central-force problem is

$$H = \frac{p_r^2}{2\mu} + \frac{\ell^2}{2\mu r^2} + U(r),$$

where $p_r = \mu \dot{r}$ is the radial canonical momentum. Since energy is also conserved, we solve

$$E = \frac{\mu \dot{r}^2}{2} + \frac{\ell^2}{2\mu r^2} + U(r) = \frac{\mu \dot{r}^2}{2} + V(r),$$

as

$$\dot{r} = \pm \sqrt{\frac{2}{\mu} [E - V(r)]}, \quad (5)$$

where $V(r)$ is known as the *effective* potential and the sign \pm depends on initial conditions. This equation can then be used with Eq. (2) to yield

$$d\theta = \frac{\ell}{\mu r^2} dt = \frac{\ell}{\mu r^2} \frac{dr}{\dot{r}} = \frac{-ds}{\sqrt{\epsilon - 2\bar{U}(s) - s^2}}, \quad (6)$$

where $\epsilon = 2\mu E/\ell^2$, or

$$s'(\theta) = \pm \sqrt{\epsilon - 2\bar{U}(s) - s^2}. \quad (7)$$

We readily check that this equation is a proper solution of the radial force equation (4) since

$$s'' = \frac{s' [d\bar{U}/ds + s]}{\sqrt{\epsilon - 2\bar{U}(s) - s^2}} = -\frac{d\bar{U}}{ds} - s$$

is indeed identical to Eq. (4). Hence, for a given central-force potential $U(r)$, we can solve for $r(\theta) = 1/s(\theta)$ by integrating

$$\theta(s) = - \int_{s_0}^s \frac{d\sigma}{\sqrt{\epsilon - 2\bar{U}(\sigma) - \sigma^2}}, \quad (8)$$

where s_0 defines $\theta(s_0) = 0$, and performing the inversion $\theta(s) \rightarrow s(\theta)$.

2.3 Turning Points

Eq. (7) yields the following energy equation

$$E = \frac{\mu}{2} \dot{r}^2 + \frac{\ell^2}{2\mu r^2} + U(r) = \frac{\ell^2}{2\mu} \left[(s')^2 + s^2 + 2\bar{U}(s) \right],$$

where $s' = -\mu\dot{r}/\ell = -p_r/p_\theta$. *Turning* points are those special values of r_n (or s_n) ($n = 1, 2, \dots$) for which

$$E = U(r_n) + \frac{\ell^2}{2\mu r_n^2} = \frac{\ell^2}{\mu} \left[\bar{U}(s_n) + \frac{s_n^2}{2} \right],$$

i.e., \dot{r} (or s') vanishes at these points. If two non-vanishing turning points $r_2 < r_1 < \infty$ (or $0 < s_1 < s_2$) exist, the motion is said to be *bounded* in the interval $r_2 < r < r_1$ (or $s_1 < s < s_2$), otherwise the motion is *unbounded*.

3 Kepler Problem

We now solve the Kepler problem where $U(r) = -k/r$, where k is a constant, so that $\bar{U}(s) = -s_0s$, where $s_0 = \mu k/\ell^2$. The turning points for the Kepler problem are solutions of the quadratic equation

$$s^2 - 2s_0s - \epsilon = 0,$$

which can be written as $s = s_0 \pm \sqrt{s_0^2 + \epsilon}$

$$s_1 = s_0(1 - e) \quad \text{and} \quad s_2 = s_0(1 + e),$$

where

$$e = \sqrt{1 + \epsilon/s_0^2} = \sqrt{1 + 2E\ell^2/\mu k^2}.$$

We note that motion is bounded when $E < 0$ ($0 < e < 1$) and unbounded when $E \geq 0$ ($e > 1$).

3.1 Bounded Keplerian Orbits

We will now look at the bounded case ($e < 1$). We define $\theta(s_2) = 0$, so that for the Kepler problem, Eq. (8) becomes

$$\theta(s) = - \int_{s_0(1+e)}^s \frac{d\sigma}{\sqrt{s_0^2 e^2 - (\sigma - s_0)^2}}, \quad (9)$$

which can easily be integrated by using the identity

$$\frac{-dx}{\sqrt{a^2 - x^2}} = d \arccos\left(\frac{x}{a}\right),$$

so that Eq. (9) yields

$$\theta(s) = \arccos\left(\frac{s - s_0}{s_0 e}\right).$$

This equation can easily be inverted to yield

$$s(\theta) = s_0(1 + e \cos \theta). \quad (10)$$

We can readily check that this solution also satisfies the radial force equation (4).

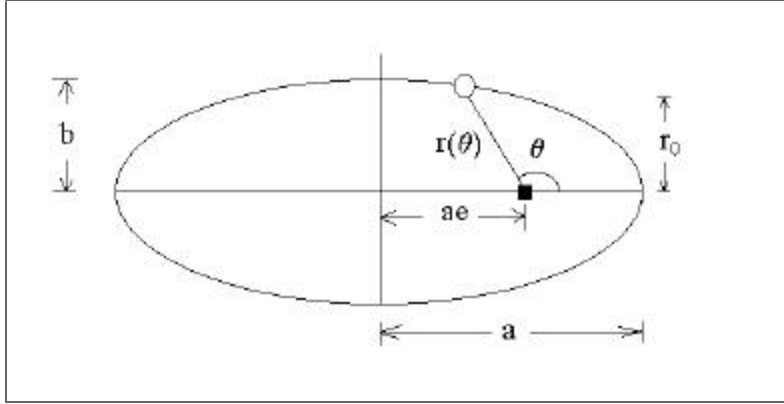
3.1.1 Kepler's First Law

The solution for $r(\theta)$ is now trivially obtained from $s(\theta)$ as

$$r(\theta) = \frac{r_0}{1 + e \cos \theta}, \quad (11)$$

where $r_0 = 1/s_0$ denotes the position of the minimum of the *effective* potential

$$V'(r_0) = \frac{1}{r_0^2} \left(k - \frac{\ell^2}{\mu r_0} \right) = 0.$$



Eq. (11) generates an ellipse of major radius

$$a = \frac{r_0}{1 - e^2} = \frac{k}{2|E|}$$

and minor radius

$$b = a\sqrt{1 - e^2} = \sqrt{\frac{\ell^2}{2\mu|E|}}$$

and, therefore, yields Kepler's First Law.

3.1.2 Kepler's Second Law

Using Eq. (2), we find

$$dt = \frac{\mu}{\ell} r^2 d\theta = \frac{2\mu}{\ell} dA(\theta),$$

where $dA(\theta) = \int r dr d\theta = \frac{1}{2} [r(\theta)]^2 d\theta$ denotes an infinitesimal area swept by $d\theta$. When integrated, this relation yields Kepler's Second law

$$\Delta t = \frac{2\mu}{\ell} \Delta A, \tag{12}$$

i.e., equal areas are swept in equal times since μ and ℓ are constants.

3.1.3 Kepler's Third Law

The orbital period T of a bound system is defined as

$$T = \int_0^{2\pi} \frac{d\theta}{\dot{\theta}} = \frac{\mu}{\ell} \int_0^{2\pi} r^2 d\theta = \frac{2\mu}{\ell} A = \frac{2\pi\mu}{\ell} ab$$

where $A = \pi ab$ denotes the area of an ellipse with major radius a and minor radius b . Using the expressions for a and b found above, we find

$$T = \frac{2\pi\mu}{\ell} \cdot \frac{k}{2|E|} \cdot \sqrt{\frac{\ell^2}{2\mu|E|}} = 2\pi \sqrt{\frac{\mu k^2}{(2|E|)^3}}.$$

If we now substitute the expression for $a = k/2|E|$ and square both sides of this equation, we obtain Kepler's Third Law

$$T^2 = \frac{(2\pi)^2\mu}{k} a^3. \quad (13)$$

Note that in Newtonian gravitational theory, $k/\mu = G(m_1 + m_2)$; although Kepler's Third Law states that T^2/a^3 is a constant for all planets in the solar system, we find that this is only an approximation that holds for $m_1 \gg m_2$.

3.2 Unbounded Keplerian Orbits

We now look at the case where the total energy is positive or zero (i.e., $e \geq 1$). Eq. (11) yields $r(1 + e \cos \theta) = r_0$ or

$$\left(\sqrt{e^2 - 1} x - \frac{er_0}{\sqrt{e^2 - 1}} \right)^2 - y^2 = \frac{r_0^2}{e^2 - 1}.$$

For $e = 1$, the particle orbit is a parabola $x = (r_0^2 - y^2)/2r_0$, with distance of closest approach at $x(0) = r_0/2$, while for $e > 1$, the particle orbit is a hyperbola.

3.3 Laplace-Runge-Lenz Vector

Let us now investigate an additional constant of the motion for the Kepler problem. First, we consider the time derivative of the vector $\mathbf{p} \times \mathbf{L}$, where the linear momentum \mathbf{p} and angular momentum \mathbf{L} are

$$\mathbf{p} = \mu (\dot{r} \hat{r} + r\dot{\theta} \hat{\theta}) \quad \text{and} \quad \mathbf{L} = \ell \hat{z} = \mu r^2 \dot{\theta} \hat{z}.$$

The time derivative of the linear momentum is $\dot{\mathbf{p}} = -\nabla U(r) = -U'(r) \hat{r}$ while the angular momentum $\mathbf{L} = \mathbf{r} \times \mathbf{p}$ is itself a constant of the motion so that

$$\begin{aligned} \frac{d}{dt}(\mathbf{p} \times \mathbf{L}) &= \frac{d\mathbf{p}}{dt} \times \mathbf{L} = -\mu \dot{r} \cdot \nabla U \mathbf{r} + \mu \mathbf{r} \cdot \nabla U \dot{\mathbf{r}} \\ &= -\frac{d}{dt}(\mu U \mathbf{r}) + \mu (\mathbf{r} \cdot \nabla U + U) \dot{\mathbf{r}}, \end{aligned}$$

hence the vector $\mathbf{A} = \mathbf{p} \times \mathbf{L} + \mu U(r) \mathbf{r}$ is a constant of the motion if the potential $U(r)$ satisfies the condition $\mathbf{r} \cdot \nabla U(r) + U(r) = 0$. For the Kepler problem, with central potential $U(r) = -k/r$, the Laplace-Runge-Lenz (LRL) vector

$$\mathbf{A} = \mathbf{p} \times \mathbf{L} - k\mu \hat{\mathbf{r}} = \left(\frac{\ell^2}{r} - k\mu \right) \hat{\mathbf{r}} - \ell \mu \dot{r} \hat{\boldsymbol{\theta}}$$

is, therefore, a constant of the motion since $\mathbf{r} \cdot \nabla U = -U$.

Since the vector \mathbf{A} is constant in both magnitude and direction, we choose its direction to be along the x -axis and its amplitude is determined at the distance of closest approach $r_{min} = r_0/(1 + e)$ and we can easily show that $\mathbf{A} \cdot \hat{\mathbf{r}} = A \cos \theta$ leads to the Kepler solution

$$r(\theta) = \frac{r_0}{1 + e \cos \theta},$$

where $r_0 = \ell^2/k\mu$ and $e = A/k\mu$.

Note that if the Keplerian orbital motion is perturbed by the introduction of an additional potential term $\delta U(r)$, we can show that the LRL vector is no longer conserved (i.e., $d\mathbf{A}/dt \neq 0$) and that the direction of the Keplerian elliptical orbit precesses with a precession frequency

$$\omega_p(\theta) = \hat{\mathbf{z}} \cdot \frac{\mathbf{A}}{A^2} \times \frac{d\mathbf{A}}{dt},$$

where the unperturbed Kepler solution $r(\theta)$ is to be used.

4 Isotropic Simple Harmonic Oscillator

We now investigate the case when the central potential is of the form

$$U(r) = \frac{k}{2} r^2 \quad \rightarrow \quad \bar{U}(s) = \frac{\mu k}{2\ell^2 s^2}. \quad (14)$$

The turning points for this problem are expressed as

$$r_1 = r_0 \left(\frac{1 - e}{1 + e} \right)^{\frac{1}{4}} = \frac{1}{s_2} \quad \text{and} \quad r_2 = r_0 \left(\frac{1 + e}{1 - e} \right)^{\frac{1}{4}} = \frac{1}{s_1},$$

where $r_0 = (\ell^2/\mu k)^{1/4} = 1/s_0$ is the radial position at which the effective potential has a minimum $V_0 = k r_0^2$ and

$$e = \sqrt{1 - \left(\frac{k r_0^2}{E} \right)^2}.$$

Here, we see that orbits exist and are always bounded for $E > V_0$ (and thus $0 < e < 1$). Next, using the change of coordinate $q = s^2$ in Eq. (8), we obtain

$$\theta = \frac{-1}{2} \int_{q_2}^q \frac{dq}{\sqrt{(\mu/\ell^2) [2E q - k] - q^2}}, \quad (15)$$

where $q_2 = (1 + e)\mu E/\ell^2$. We now substitute $q(\varphi) = (1 + e \cos \varphi)\mu E/\ell^2$ in Eq. (15) to obtain

$$\theta = \frac{1}{2} \arccos \left[\frac{1}{e} \left(\frac{\ell^2 q}{\mu E} - 1 \right) \right]$$

or

$$r(\theta) = \frac{r_0 \sqrt{1 - e^2}}{\sqrt{1 + e \cos 2\theta}}. \quad (16)$$

This equation describes the ellipse

$$\frac{x^2}{(1 - e)} + \frac{y^2}{(1 + e)} = r_0^2$$

of semi-major axis $a = r_0 \sqrt{1 + e}$ and semi-minor axis $b = r_0 \sqrt{1 - e}$. Lastly, we note that one revolution along the orbit $r(\theta)$ corresponds to an angular period of π , i.e., $r(\theta + \pi) = r(\theta)$, and not 2π as found in the Kepler problem. The area of the ellipse $A = \pi ab = \pi(\ell^2/\mu E)$ while the period is

$$T(E, \ell) = \int_0^\pi \frac{d\theta}{\dot{\theta}} = \frac{\mu A}{\ell} = \frac{\pi \ell}{E}.$$

If we introduce the angular frequency $\omega = \pi/T$, then we find the important relation between energy and angular momentum $E = \ell \omega$.